

中国民用航空总局
GENERAL ADMINISTRATION OF CIVIL AVIATION OF CHINA

编号/No.: VTC131A 修订/ Rev.: 0 型号/Type: 182T, T182T 日期/Date: 26/07/2002

型号认可证数据单编号: VTC131A
Validation Type Certificate Data Sheet No. VTC131A

本数据单, 作为型号认可证编号: VTC131A 的一部分, 规定了一些条件和限制, 在此条件和限制下, 获得本型号认可证的产品满足下述“认可审定基础”段落给出的适航要求。
This data sheet which is a part of Validation Type Certificate No. VTC131A prescribes conditions and limitations under which the product for which the Validation Type Certificate was issued meets the airworthiness requirements indicated in the paragraph “Validation Basis”.

美国 FAA 型号合格证 No.3A13、数据单第 59 版, 批准日期: 2001 年 12 月 12 日, 其中
182T (正常类): 2001 年 2 月 23 日 FAA 批准
T182T (正常类): 2001 年 2 月 23 日 FAA 批准
Type Certificate No.3A13, Data Sheet Revision 59, Approved by FAA on December 12, 2001
Model 182T(Normal Category): Approved by FAA on 23 February 2001.
Model T182T(Normal Category): Approved by FAA on 23 February 2001.

型号合格证持有人: 赛斯纳飞机公司
美国堪萨斯州威奇塔城邮政信箱7704号
邮政编码: 67277
Type Certificate Holder: Cessna Aircraft Company
P. O. Box 7704
Wichita, Kansas 67277

I-182T 型飞机, Skylane, 4 座、封闭座舱、陆上型单翼飞机 (正常类)
I - Model 182T, Skylane, 4PCLM (Normal Category)

发动机 莱康明公司制造的 IO-540-AB1A5, 额定 230 马力
Engine Lycoming IO-540-AB1A5, Rated 230 Horsepower

燃油 最低航空汽油等级 100/100LL
Fuel 100/100LL minimum grade aviation gasoline

发动机限制 在所有使用情况下, 2400 转/分钟
Engine Limits For all operations, 2400 RPM

螺旋桨 麦可利公司制造的恒速螺旋桨
(a) 螺旋桨: B3D36C431/80VSA-1 (三叶)
(b) 直径: 不超过79英寸, 不低于77.5英寸
(c) 在站位30英寸处的桨距设定: 低距14.9°, 高距31.7°
(d) 麦可利桨毂: D-7261-2
(e) 麦可利调速器DC290D1/T8
Propeller McCauley Constant Speed
(a) Propeller: B3D36C431/80VSA-1 (3 blades)
(b) Diameter: not over 79 in., not under 77.5 in.
(c) Pitch settings at 30 in. sta.: Low 14.9°, High 31.7°
(d) McCauley Spinner: D-7261-2
(e) McCauley Governor DC290D1/T8

螺旋桨限制 在全推力下的静态转速: 不超过2400转/分钟, 不低于2300转/分钟

Propeller Limits: Static RPM at full throttle: Not over 2400; Not Under 2300

空速限制

机动速度	110节 (指示空速)	(108节 (校正空速))
最大结构巡航速度	140节 (指示空速)	(138节 (校正空速))
不可超越速度	175节 (指示空速)	(171节 (校正空速))
襟翼展态速度	100节 (指示空速)	(99节 (校正空速))

Airspeed Limits

Maneuvering	110 Knots IAS	(108 Knots CAS)
Max. Structural Cruising	140 Knots IAS	(138 Knots CAS)
Never Exceed	175 Knots IAS	(171 Knots CAS)
Flaps Extended	100 Knots IAS	(99 Knots CAS)

重心范围

正常类

(1) 后限: 在不大于3100磅时, 基准之后的46英寸处

(2) 前限: 在3100磅时, 基准之后的40.9英寸处; 在2700磅时, 基准之后的35.5英寸处; 在上述两者重量之间的重量时, 线性变化; 不大于2250磅时, 基准之后的33.0英寸处。

CG Range:

Normal Category

(1) Aft Limits: 46 inches aft of datum at 3100 lbs. or less.

(2) Forward Limits: Linear variation from 40.9 inches aft of datum at 3100 pounds to 35.5 inches aft of datum at 2700 pounds; 33.0 inches aft of datum at 2250 pounds or less.

空机重心:

无

Empty Wt. C.G. Range:

None

基准:

防火墙前表面的较低部位

Reference Datum:

Lower portion of front face of Firewall

平均气动弦:

58.8英寸; 平均气动弦前缘为基准之后的25.98英寸处

MAC:

58.8 inches; Leading edge of MAC 25.98 inches aft of datum

调水平措施

尾锥左侧、基准之后的139.65英寸和171.65英寸处

Leveling Means:

Left side of Tailcone at 139.65 inches and 171.65 inches aft of datum

最大重量:

正常类

最大停机坪重量 3110磅

最大起飞重量 3100磅

最大着陆重量 2950磅

Maximum Weights:

Normal Category

Maximum Ramp: 3110 pounds

Maximum Takeoff: 3100 pounds

Maximum Landing: 2950 pounds

座位数量

4 (2个座位在基准之后的32.0英寸至50.0英寸之间; 2个座椅在基准之后的74.0英寸)

No. of Seats:

4 (2 at 32.0 to 50.0 inches aft of datum; 2 at 74.0 inches aft of datum)

最大行李重量:

120磅在基准之后的82.0至109.0英寸之间

80磅在基准之后的109.0至134.0英寸之间

(行李舱位置的最大累积承重量为200磅)

Maximum Baggage:

120 pounds at 82.0 to 109.0 mchs aft of datum

80 pounds at 109.0 to 134.0 inches aft of datum

(Max. combined weight capacity for baggage area is 200 pounds)

燃油容量

总计92加仑; 可用燃油87加仑

Fuel Capacity (Gal.):

92 gal. total; 87 gal. Usable

(两个46加仑的机翼油箱, 位于基准之后的46.5英寸处)

(Two 46 gallon tanks in wings at 46.5 inches aft of datum)

关于不可用燃油, 见注释1

See NOTE 1 for data on unusable fuel.

滑油容量

9.0夸脱, 基准之前的14.8英寸处; 可用滑油5.0夸脱

Oil Capacity:

9.0 quarts at 14.8 inches forward of datum; 5.0 quarts usable

操纵面位移

机翼襟翼:		向下38° +0°, -1°
升降舵调整片:	向上24°±2°	向下15°±1°
副翼:	向上20°±2°	向下15°±2°
升降舵:	向上28°±1°	向下21°±1°
(相对于水平安定面)		
方向舵:	向右24° +0°, -1°	向左24° +0°, -1°
	(平行于水线)	
	向右27°13' +0°, -1°	向左27°13' +0°, -1°
	(垂直于铰链线)	

Control Surface Movements:

Wing Flaps:		Down 38° +0°, -1°
Elevator Tab:	Up 24°±2°	Down 15°±1°
Ailerons:	Up 20°±2°	Down 15°±2°
Elevator:	Up 28°±1°	Down 21°±1°
(Relative to stabilizer)		
Rudder:	Right: 24° +0°, -1°	Left: 24° +0°, -1°
	(Parallel to 0.00 W.L.)	
	Right: 27°13' +0°, -1°	Left: 27°13' +0°, -1°
	(Perpendicular to hinge line)	

制造人序列号 18280945及其以后
Serial Nos. Eligible 18280945 and On

182T 型相关的数据**Data Pertinent to Model 182T:**

FAA审定基础

1965年2月1日生效的联邦航空条例，修正案为23-1至23-6,但以下条款例外:

FAR 23.423; 23.611; 23.619; 23.623; 23.689; 23.775; 23.871; 23.1323和 23.1563的修正案为23-7; FAR 23.807 和 23.1524的修正案为23-10; FAR 23.507; 23.771; 23.853(a),(b),(c)和23.1365的修正案为23-14; FAR 23.951的修正案为23-15; FAR 23.607; 23.675; 23.685; 23.733; 23.787; 23.1309和23.1322的修正案为23-17; FAR 23.1301的修正案为23-20; FAR 23.1353和23.1559的修正案为23-21; FAR 23.603; 23.605; 23.613; 23.1329和23.1545的修正案为23-23; FAR 23.441和23.1549的修正案为23-28; FAR 23.779和23.781的修正案为23-33; FAR 23.1; 23.51 和23.561的修正案为23-34; FAR 23.301; 23.331; 23.351; 23.427; 23.677; 23.701; 23.735和23.831的修正案为23-42; FAR 23.961; 23.1093; 23.1143(g); 23.1147(b); 23.1303; 23.1357; 23.1361和23.1385的修正案为23-43; FAR 23.562(a), 23.562(b)2, 23.562(c)1, 23.562(c)2, 23.562(c)3和23.562(c)4的修正案为23-44; FAR 23.33; 23.53; 23.305; 23.321; 23.485; 23.621; 23.655和23.731的修正案为23-45。

1969年12月1日生效的FAR 36，修正案为36-1至36-21。

FAA Certification Basis

Part 23 of the Federal Aviation Regulations effective February 1, 1965, as amended by 23-1 through 23-6, except as follows:

FAR 23.423; 23.611; 23.619; 23.623; 23.689; 23.775; 23.871; 23.1323; and 23.1563 as amended by Amendment 23-7. FAR 23.807 and 23.1524 as amended by Amendment 23-10. FAR 23.507; 23.771; 23.853(a),(b) and (c); and 23.1365 as amended by Amendment 23-14. FAR 23.951 as amended by Amendment 23-15. FAR 23.607; 23.675; 23.685; 23.733; 23.787; 23.1309 and 23.1322 as amended by Amendment 23-17. FAR 23.1301 as amended by Amendment 23-20. FAR 23.1353; and 23.1559 as amended by Amendment 23-21. FAR 23.603; 23.605; 23.613; 23.1329 and 23.1545 as amended by Amendment 23-23. FAR 23.441 and 23.1549 as amended by Amendment 23-28. FAR 23.779 and 23.781 as amended by Amendment 23-33. FAR 23.1; 23.51 and 23.561 as amended by Amendment 23-34. FAR 23.301; 23.331; 23.351; 23.427; 23.677; 23.701; 23.735; and 23.831 as amended by Amendment 23-42. FAR 23.961; 23.1093; 23.1143(g); 23.1147(b); 23.1303; 23.1357; 23.1361 and 23.1385 as amended by Amendment 23-43. FAR 23.562(a), 23.562(b)2, 23.562(c)1, 23.562(c)2, 23.562(c)3, and 23.562(c)4 as amended by Amendment 23-44. FAR 23.33; 23.53; 23.305; 23.321; 23.485; 23.621; 23.655 and 23.731 as amended by Amendment 23-45.

FAR 36 dated December 1, 1969, as amended by Amendments 36-1 through 36-21.

等效安全条款:

- (1) 进气系统防冰
(2) 油门控制

FAR §1093; 参考2000年12月19日的FAA信函

FAR § 23.1143(g); 参考2000年12月19日的FAA信函

- (3) 油气混合控制 FAR § 23.1147(b); 参考2000年12月19日的FAA信函
(4) 防撞灯 FAR § 23.1401(d)

Equivalent Safety Items:

- (1) Induction System Icing Protection FAR §1093; Refer to FAA letter dated 12/19/00
(2) Throttle Control FAR § 23.1143(g); Refer to FAA letter dated 12/19/00
(3) Mixture Control FAR § 23.1147(b); Refer to FAA letter dated 12/19/00
(4) Anti-collision Lights FAR § 23.1401(d)

CAAC 认可审定基础 CAAC 接受 FAA 的审定基础为认可审定基础
CAAC Validation Basis CAAC Accept FAA certification basis as validation basis

生产依据 生产许可证编号 PC-4
Production Basis Production Certificate No. PC-4

设备 认可审定基础规定的必要的设备必须安装在飞机上供适航审定
Equipment The basic required equipment as prescribed in the Validation Basis must be installed in the airplane for
certification.

注释 1: 重量和平衡
制造人序列号: 18280945 和及其之后的 (182T)
审定的基本空重及其相应的重心位置必须包括不可用燃油 24 磅 (基准之后的 48 英寸处) 和全部滑油 16.2 磅 (基准之前的 14.8 英寸处)。

NOTE 1: Weight and Balance:
Serial Nos. 18280945 and On; (Model 182T)
The certificated basic empty weight and corresponding center of gravity location must include unusable fuel of 24lbs. at 48 inches aft of datum, and full oil of 16.2 lb. at 14.8 inches forward of datum.

注释 2: 《FAA 批准的飞机飞行手册(AFM)》: 编号 182T PHUS00 (或 FAA 后续批准的修订) 适用于 182 T 型飞机。182 T 型飞机必须依照相应的飞行手册运行。要求的标牌包括在飞机飞行手册中。

NOTE 2: FAA Approved Airplane Flight Manual (AFM): Part number 182TPHUS00 (or later FAA approved revision)is applicable to the Model 182T.The Airplane must be operated according to the appropriate AFM. Required placards are included in the AFM.

注释 3: 汽缸头温度(CHT)测量探头必须安装在 3 号汽缸头上。

NOTE 3: The CHT probe must be installed on Head #3.

I-T182T 型飞机, Skylane, 4 座、封闭座舱、陆上型单翼飞机 (正常类)**I - Model T182T, Skylane, 4PCLM (Normal Category)**

发动机 莱康明公司制造的 TIO-540-AK1A, 额定 235 马力
Engine Lycoming TIO-540-AK1A, Rated 235 Horsepower

燃油 最低航空汽油等级 100/100LL
Fuel 100/100LL minimum grade aviation gasoline

发动机限制 在所有使用情况下, 2400 转/分钟
Engine Limits For all operations, 2400 RPM

螺旋桨 麦可利公司制造的恒速螺旋桨
(a) 螺旋桨: B3D36C442/80VSB-1 (三叶)
(b) 直径: 不超过 79 英寸, 不低于 77.5 英寸
(c) 在站位 30 英寸处的桨距设定: 低距 15.30, 高距 35.40
(d) 麦可利桨毂: D-7261-2
(e) 麦可利调速器 DC290D1/T8
Propeller McCauley Constant Speed
(a) Propeller: B3D36C442/80VSB-1 (3 blades)
(b) Diameter: not over 79 in., not under 77.5 in.
(c) Pitch settings at 30 in. sta.: Low 15.3°, High 35.4°
(d) McCauley Spinner: D-7261-2
(e) McCauley Governor DC290D1/T8

螺旋桨限制	在全推力下的静态转速: 不超过 2400 转/分钟, 不低于 2300 转/分钟		
Propeller Limits:	Static RPM at full throttle: Not over 2400, Not Under 2300		
空速限制			
	机动速度	110 节 (指示空速)	(110 节 (校正空速))
	最大结构巡航速度	140 节 (指示空速)	(137 节 (校正空速))
	不可超越速度	175 节 (指示空速)	(170 节 (校正空速))
	襟翼展态速度	100 节 (指示空速)	(100 节 (校正空速))
Airspeed Limits			
	Maneuvering	110 Knots IAS	(110 Knots CAS)
	Max. Structural Cruising	140 Knots IAS	(137 Knots CAS)
	Never Exceed	175 Knots IAS	(170 Knots CAS)
	Flaps Extended	100 Knots IAS	(100 Knots CAS)
重心范围	正常类		
	(1) 后限: 在不大于 3100 磅时, 基准之后的 46 英寸处		
	(2) 前限: 在 3100 磅时, 基准之后的 40.9 英寸处; 在 2700 磅时, 基准之后的 35.5 英寸处; 在上述两者重量之间的重量时, 线性变化; 不大于 2250 磅时, 基准之后的 33.0 英寸处。		
CG Range:	Normal Category		
	(1) Aft Limits: 46 inches aft of datum at 3100 lbs. or less.		
	(2) Forward Limits: Linear variation from 40.9 inches aft of datum at 3100 pounds to 35.5 inches aft of datum at 2700 pounds; 33.0 inches aft of datum at 2250 pounds or less.		
空机重心:	无		
Empty Wt. C.G. Range:	None		
基准:	防火墙前表面的较低部位		
Reference Datum:	Lower portion of front face of Firewall		
平均气动弦:	58.8 英寸; 平均气动弦前缘为基准之后的 25.98 英寸处		
MAC:	58.8 inches; Leading edge of MAC 25.98 inches aft of datum		
调水平措施	尾锥左侧、基准之后的 139.65 英寸和 171.65 英寸处		
Leveling Means:	Left side of Tailcone at 139.65 inches and 171.65 inches aft of datum		
最大重量:	正常类		
	最大停机坪重量	3110 磅	
	最大起飞重量	3100 磅	
	最大着陆重量	2950 磅	
Maximum Weights:	Normal Category		
	Maximum Ramp:	3110 pounds	
	Maximum Takeoff:	3100 pounds	
	Maximum Landing:	2950 pounds	
座位数量	4 (2 个座位在基准之后的 32.0 英寸至 50.0 英寸之间; 2 个座椅在基准之后的 74.0 英寸)		
No. of Seats:	4 (2 at 32.0 to 50.0 inches aft of datum; 2 at 74.0 inches aft of datum)		
最大行李重量:	120 磅在基准之后的 82.0 至 109.0 英寸之间		
	80 磅在基准之后的 109.0 至 134.0 英寸之间		
	(行李舱位置的最大累积承重量为 200 磅)		
Maximum Baggage:	120 pounds at 82.0 to 109.0 inches aft of datum		
	80 pounds at 109.0 to 134.0 inches aft of datum		
	(Max. combined weight capacity for baggage area is 200 pounds)		
燃油容量	总计 92 加仑; 可用燃油 87 加仑		
	(两个 46 加仑的机翼油箱, 位于基准之后的 46.5 英寸处)		
	关于不可用燃油, 见注释 1		
Fuel Capacity (Gal.):	92 gal. total; 87 gal. Usable		
	(Two 46 gallon tanks in wings at 46.5 inches aft of datum)		
	See NOTE 1 for data on unusable fuel.		
滑油容量	9.0 夸脱, 基准之前的 14.8 英寸处; 可用滑油 5 夸脱		

Oil Capacity : 9.0 quarts at 14.8 inches forward of datum; 5 quarts usable

操纵面位移

机翼襟翼:		向下 38° +0°, -1°
升降舵调整片:	向上 24°±2°	向下 15°±1°
副翼:	向上 20°±2°	向下 15°±2°
升降舵:	向上 28°±1°	向下 21°±1°
(相对于水平安定面)		
方向舵:	向右 24° +0°, -1°	向左 24° +0°, -1°
(平行于水线)		
	向右 27°13' +0°, -1°	向左 27°13' +0°, -1°
(垂直于铰链线)		

Control Surface Movements:

Wing Flaps:		Down 38° +0°, -1°
Elevator Tab:	Up 24°±2°	Down 15°±1°
Ailerons:	Up 20°±2°	Down 15°±2°
Elevator:	Up 28°±1°	Down 21°±1°
(Relative to stabilizer)		
Rudder:	Right: 24° +0°, -1°	Left: 24° +0°, -1°
(Parallel to 0.00 W.L.)		
	Right: 27°13' +0°, -1°	Left: 27°13' +0°, -1°
(Perpendicular to hinge line)		

制造人序列号 T18208001 及其以后

Serial Nos. Eligible T18208001 and On

T182T 型相关的数据

Data Pertinent to Model T182T:

FAA 审定基础

1965 年 2 月 1 日生效的联邦航空条例, 修正案为 23-1 至 23-6, 但以下条款例外:

FAR 23.423; 23.611; 23.619; 23.623; 23.689; 23.775; 23.871; 23.1323 和 23.1563 的修正案为 23-7; FAR 23.807 和 23.1524 的修正案为 23-10; FAR 23.507; 23.771; 23.853(a),(b),(c) 和 23.1365 的修正案为 23-14; FAR 23.951 的修正案为 23-15; FAR 23.607; 23.675; 23.685; 23.733; 23.787; 23.1309 和 23.1322 的修正案为 23-17; FAR 23.1301 的修正案为 23-20; FAR 23.1353 和 23.1559 的修正案为 23-21; FAR 23.603; 23.605; 23.613; 23.1329 和 23.1545 的修正案为 23-23; FAR 23.441 和 23.1549 的修正案为 23-28; FAR 23.779 和 23.781 的修正案为 23-33; FAR 23.1; 23.51 和 23.561 的修正案为 23-34; FAR 23.301; 23.331; 23.351; 23.427; 23.677; 23.701; 23.735 和 23.831 的修正案为 23-42; FAR 23.961; 23.1093; 23.1143(g); 23.1147(b); 23.1303; 23.1357; 23.1361 和 23.1385 的修正案为 23-43; FAR 23.562(a), 23.562(b)2, 23.562(c)1, 23.562(c)2, 23.562(c)3 和 23.562(c)4 的修正案为 23-44; FAR 23.33; 23.53; 23.305; 23.321; 23.485; 23.621; 23.655 和 23.731 的修正案为 23-45。

1969 年 12 月 1 日生效的 FAR 36, 修正案为 36-1 至 36-22。

FAA Certification Basis

Part 23 of the Federal Aviation Regulations effective February 1, 1965, as amended by 23-1 through 23-6, except as follows:

FAR 23.423; 23.611; 23.619; 23.623; 23.689; 23.775; 23.871; 23.1323; and 23.1563 as amended by Amendment 23-7. FAR 23.807 and 23.1524 as amended by Amendment 23-10. FAR 23.507; 23.771; 23.853(a),(b) and (c); and 23.1365 as amended by Amendment 23-14. FAR 23.951 as amended by Amendment 23-15. FAR 23.607; 23.675; 23.685; 23.733; 23.787; 23.1309 and 23.1322 as amended by Amendment 23-17. FAR 23.1301 as amended by Amendment 23-20. FAR 23.1353; and 23.1559 as amended by Amendment 23-21. FAR 23.603; 23.605; 23.613; 23.1329 and 23.1545 as amended by Amendment 23-23. FAR 23.441 and 23.1549 as amended by Amendment 23-28. FAR 23.779 and 23.781 as amended by Amendment 23-33. FAR 23.1; 23.51 and 23.561 as amended by Amendment 23-34. FAR 23.301; 23.331; 23.351; 23.427; 23.677; 23.701; 23.735; and 23.831 as amended by Amendment 23-42. FAR 23.961; 23.1093; 23.1143(g); 23.1147(b); 23.1303; 23.1357; 23.1361 and 23.1385 as amended by Amendment 23-43. FAR 23.562(a), 23.562(b)2, 23.562(c)1, 23.562(c)2, 23.562(c)3, and 23.562(c)4 as amended by Amendment 23-44. FAR 23.33; 23.53; 23.305; 23.321; 23.485; 23.621; 23.655 and 23.731 as amended by Amendment 23-45.

FAR 36 dated December 1, 1969, as amended by Amendments 36-1 through 36-22.

等效安全条款:

- | | |
|------------|---|
| (1) 油门控制 | FAR § 23.1143(g); 参考 2000 年 12 月 19 日的 FAA 信函 |
| (2) 油气混合控制 | FAR § 23.1147(b); 参考 2000 年 12 月 19 日的 FAA 信函 |
| (3) 防撞灯 | FAR § 23.1401(d) |

Equivalent Safety Items:

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|---------------------------|--|
| (1) Throttle Control | FAR § 23.1143(g); Refer to FAA letter dated 12/19/00 |
| (2) Mixture Control | FAR § 23.1147(b); Refer to FAA letter dated 12/19/00 |
| (3) Anti-collision Lights | FAR § 23.1401(d) |

CAAC 认可审定基础	CAAC 接受 FAA 的审定基础为认可审定基础
CAAC Validation Basis	CAAC Accept FAA certification basis as validation basis

生产依据	生产许可证编号 PC-4
Production Basis	Production Certificate No. PC-4

设备	认可审定基础规定的必要的设备必须安装在飞机上供适航审定
Equipment	The basic required equipment as prescribed in the Validation Basis must be installed in the airplane for certification.

注释 1: 重量和平衡
制造人序列号: T18208001 和及其之后的 (T182T)
审定的基本空重及其相应的重心位置必须包括不可用燃油 30 磅 (基准之后的 48.0 英寸处) 和全部滑油 16.2 磅 (基准之前的 14.8 英寸处)。

NOTE 1: Weight and Balance:
Serial Nos. T18208001 and On; (Model T182T)
The certificated basic empty weight and corresponding center of gravity location must include unusable fuel of 30lbs. at 48.0 inches aft of datum, and full oil of 16.2 lb. at 14.8 inches forward of datum.

注释 2: 《FAA 批准的飞机飞行手册 (AFM)》: 编号 T182T PHUS00 (或 FAA 后续批准的修订) 适用于 T182 T 型飞机。T182 T 型飞机必须依照相应的飞行手册运行。要求的标牌包括在飞机飞行手册中。

NOTE 2: FAA Approved Airplane Flight Manual (AFM): Part number T182TPHUS00 (or later FAA approved revision) is applicable to the Model T182T. The Airplane must be operated according to the appropriate AFM. Required placards are included in the AFM.

注释 3: 汽缸头温度 (CHT) 测量探头必须安装在 4 号汽缸头上。

NOTE 3: The CHT probe must be installed on Head #4.

.....结束.....

.....END.....